ANM-100D-0239

United States of America

Department of Transportation—Federal Aviation Administration

Supplemental Type Certificate

Number SH4062NM

This certificate, issued to Air Methods Corporation

certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part 29 *of the* Federal Aviation Regulations.

Original Product—Type Certificate Number H4SW

Make: Bell Helicopter Textron

Model: 412

Description of Type Design Change:

Installation of a Spectrolab searchlight in accordance with Air Methods Corporation Master Drawing List 141-9004, Revision A, dated October 19, 1989, FAA approved November 6, 1989, or later approved revision.

Limitations and Conditions:

1. FAA approved Rotorcraft Flight Manual Supplement dated November 6, 1989, or later approved revision is required.
2. This installation is not compatible with aircraft equipped with floats or hoist systems.
3. This approval should not be extended to aircraft of this model on which other previously approved modifications are incorporated unless

(See Continuation Sheet, page 3 of 3.)

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application: September 8, 1989

Date of issuance: November 6, 1989

Date reissued:

Date amended: December 21, 1990

By direction of the Administrator

RONALD F. MAY, *(Signature)* Acting Supervisor
Denver Aircraft Certification Field Office
Northwest Mountain Region, Denver, Colorado
(Title)

Any alteration of this certificate is punishable by a fine of not exceeding $1,000, or imprisonment not exceeding 3 years, or both.

This certificate may be transferred in accordance with FAR 21.47.

FAA Form 8110-2 (10-68)

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Supplemental Type Certificate

(Continuation Sheet)

Number SH4062NM

it is determined that the interrelationship between this change and any other previously approved modifications will introduce no adverse effect upon the airworthiness of the aircraft.